

# COMPARATIVE ANALYSIS OF WORKING FLUIDS IN SOLAR JOULE BRAYTON COGENERATION ENGINES

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**Rezumat.** Lucrarea include o analiză comparativă a influenței naturii agentului de lucru asupra performanțelor unui ciclu solar cogenerativ Joule Brayton de mică putere. Schema constructivă a motorului Joule Brayton a presupus sistemul: compresor centrifug – turbină centripetă – captator/concentrator de radiație solară– recuperator intern de căldură, ce poate fi utilizat pentru puteri mici. S-au considerat trei posibili agenți de lucru, aer, azot și dioxid de carbon. Pentru a obține rezultate numerice cât mai veridice, schema numerică de calcul a considerat călduri specifice variabile, rapoarte de comprimare și randamente izentropice corespunzătoare tipurilor de compresor și turbină adoptate, diferite grade de concentrare a radiației solare prin intermediul temperaturii maxime pe ciclu. Analiza numerică a evidențiat că alegerea unui agent de lucru trebuie făcută pentru fiecare aplicație în parte, în funcție și de condiții restrictive specifice diferite de cele termodinamice, cum ar fi de exemplu raportul dintre fluxul termic cerut de consumatorii de căldură și puterea furnizată impusă de consumatorii acestei utilități energetice, tehnologii de fabricație și valoarea investițiilor specifice, etc.

**Cuvinte cheie:** ciclu solar cogenerativ Joule Brayton, agenți de lucru, performanțe termodinamice, schemă logică de calcul.

**Abstract.** The paper includes a comparative analysis of the effect of the working fluid type upon the performances of a small solar cogeneration Joule Brayton cycle. The engine scheme considered the setup: radial compressor – centripetal turbine – concentrating solar power (CSP) heat exchanger – internal recovering heat exchanger – and external cogeneration heat exchanger. They were evaluated three working fluids, air, nitrogen, and carbon dioxide. The numerical code involved variable heat capacities, compression ratios and isentropic efficiencies typical for the chosen compressors and gas turbine, and various solar radiation concentrating degrees by the intermediary of the maximum temperature on the cycle. The numerical analysis emphasized that the choosing of a certain working fluid must follow also extra non-thermodynamically based real operational restrictive conditions, such as for instance the operational heat per power ratio, technological approaches and specific financial investments etc.

**Keywords:** solar cogeneration Joule Brayton cycle, working fluids, thermodynamic performances, numerical algorithm.

## 1. INTRODUCTION

The next future of energy systems must deliberate on their structure including besides classical energy systems, fueled and nuclear, new advanced and more and more widespread renewable based ones.

These new systems are to be “standardized” by more points of view, for instance:

- thermodynamic criteria like power, first and second law efficiencies,
- technological methodologies adapted to small or large power systems, open for air or closed for other working fluids, compatible materials, design for assembling at the system operational start and disassembling at the system life end,
- financial issues as the specific investment,
- life cycle assessments, and so on.

This paper tried to find out if there are major effects of the working fluid nature upon the thermodynamic performances of the small solar Joule Brayton engines.

The numerical analysis emphasized that the choosing of a certain working fluid must follow also extra, non-thermodynamically based, real operational restrictive conditions, for instance the operational heat per power ratio, technological approaches and specific financial investments etc.

## 2. MATHEMATICAL ALGORITHM

### 2.1. Hypotheses

The operational cycle is sketched in Figure 1. The successively processes are:

- 1 – 2r irreversible compression,
- 2r – 3r irreversible pre-heating by internal heat recovering,
- 3r – 4r irreversible heating by solar radiation,
- 4r – 5r irreversible expansion,
- 5r – 6r irreversible cooling by internal heat recovering,
- 6r – 7r irreversible cooling by cogeneration,
- 7r – 1 irreversible cooling by releasing out the exhaust heat.

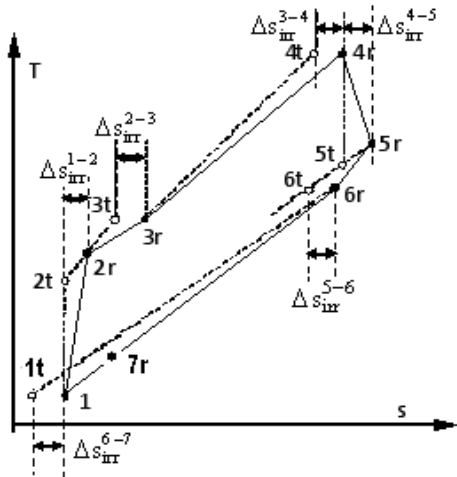


Fig. 1. The solar cogeneration Joule Brayton cycle

The adopted assumptions on the cycle are:

- the minimum temperature on the cycle  $T_1 = 293.15 \text{ K}$ ;
- the maximum temperature on the cycle  $T_{4r} = 873.15 / 1073.15 / 1273.15 / 1473.15 \text{ K}$ ;
- compression ratios  $\pi_C = p_2/p_1 = 2.5 / 3.5 / 4.5$
- isentropic efficiencies  $\eta_C = 0.8, \eta_T = 0.8$ ;
- mechanical power loss: 1%;
- effectiveness of internal heat exchanger  $\epsilon_{recov} = 0.95$ ;
- irreversible pressure drops by  $p_{3r}/p_{2r} = 0.98, p_{4r}/p_{3r} = 0.99, p_{6r}/p_{5r} = 0.970, p_{6r}/p_{7r} = 0.955, \text{ and } p_{7r}/p_1 = 0.9975$ ;
- heat capacities evaluated by polynomials of 4<sup>th</sup> order;
- ideal gas constants compiled as mean values of the difference between the constant pressure heat capacity and the constant volume heat capacity, on the range 20 – 1300 degrees Celsius.

## 2.2. Basic relations

- ◆ Heat capacities interpolated by polynomials:

$$c_p = a_0 + a_1 \cdot T + a_2 \cdot T^2 + a_3 \cdot T^3 + a_4 \cdot T^4 \quad (1)$$

$$c_v = b_0 + b_1 \cdot T + b_2 \cdot T^2 + b_3 \cdot T^3 + b_4 \cdot T^4$$

- ◆ Mean adiabatic exponents of reversible adiabatic process 1 – 2t and 4r – 5t:

$$\gamma_{12} = \frac{\int_{T_1}^{T_{2t}} c_p \cdot dT}{\int_{T_1}^{T_{2t}} c_v \cdot dT}, \quad \gamma_{45} = \frac{\int_{T_{4r}}^{T_{5t}} c_p \cdot dT}{\int_{T_{4r}}^{T_{5t}} c_v \cdot dT} \quad (2)$$

- ◆ Isentropic efficiencies of irreversible adiabatic process 1 – 2r and 4r – 5r:

$$\eta_C = \frac{\int_{T_1}^{T_{2r}} c_p \cdot dT}{\int_{T_1}^{T_{2r}} c_p \cdot dT}, \quad \eta_T = \frac{\int_{T_{4r}}^{T_{5r}} c_p \cdot dT}{\int_{T_{4r}}^{T_{5r}} c_p \cdot dT} \quad (3)$$

- ◆ Mean ideal gas constants over a temperature range, from  $T_{\min}$  to  $T_{\max}$ :

$$R = \frac{\int_{T_{\min}}^{T_{\max}} (c_p - c_v) \cdot dT}{T_{\max} - T_{\min}} \quad (4)$$

- ◆ Effectiveness of internal recovering heat exchanger, knowing that the flow heat capacity rate along the process 2r – 3t is smaller than that along the process 5r – 6t, since the specific heat capacity is larger while the temperature is larger:

$$\epsilon_{recov} = \frac{\int_{T_{2r}}^{T_{3t}} c_p \cdot dT}{\int_{T_{2r}}^{T_{5r}} c_p \cdot dT} \approx \frac{T_{3t} - T_{2r}}{T_{5r} - T_{2r}} \quad (5)$$

- ◆ Pressures and temperatures:

$$p_1 = 1 \text{ bar}, \quad p_{2t} = p_1 \cdot \pi_C, \quad p_{2r} = p_{2t}, \quad p_{3t} = p_{2r},$$

$$p_{3r} = 0.98 \cdot p_{3t}, \quad p_{4r} = 0.99 \cdot p_{3r}, \quad (6)$$

$$p_{7r} = \frac{p_1}{0.9975}, \quad p_{6r} = \frac{p_{7r}}{0.955}, \quad p_{5r} = \frac{p_{6r}}{0.97}$$

$$T_1 = 293.15 \text{ K}, \quad T_{3t} = T_{3r}, \quad T_{4t} = T_{4r}, \quad T_{6t} = T_{6r}, \quad T_{7r} = 363.15 \text{ K} \quad (7)$$

## 2.3. Basic numerical code equations

- ◆ Reversible adiabatic compression 1 – 2t:

$$T_{2t} = T_1 \cdot \left( \frac{p_{2t}}{p_1} \right)^{(\gamma_{12}-1)/\gamma_{12}} \quad (8)$$

Solving this equation by an iterative method, e.g. trial and error, it yields the temperature  $T_{2t}$ .

- ◆ Irreversible adiabatic compression 1 – 2r:

$$\eta_C \cdot \int_{T_1}^{T_{2r}} c_p \cdot dT = \int_{T_1}^{T_{2t}} c_p \cdot dT \quad (9)$$

By using also an iterative method, from this equation it results  $T_{2r}$ .

- ◆ Reversible adiabatic expansion 4r – 5t:

$$T_{5t} = T_{4r} \cdot \left( \frac{p_{5t}}{p_{4r}} \right)^{(\gamma_{45}-1)/\gamma_{45}} \quad (10)$$

From this equation it yields the temperature  $T_{5t}$ , by means of an iterative method, e.g. trial and error.

- ◆ Irreversible adiabatic expansion 4r – 5r:

$$\eta_T \cdot \int_{T_{4r}}^{T_{5t}} c_p \cdot dT = \int_{T_{4r}}^{T_{5r}} c_p \cdot dT \quad (11)$$

A similar way of solving this equation, it obtains  $T_{5r}$ .

- ◆ Internal heat recovering (3r – 4t) + (5r – 6t):

$$T_{3t} = \varepsilon_{recov} \cdot (T_{5r} - T_{2r}) + T_{2r}, T_{3r} = T_{3t} \quad (12)$$

$$\int_{T_{2r}}^{T_{3t}} c_p \cdot dT = - \int_{T_{5r}}^{T_{6t}} c_p \cdot dT, T_{6r} = T_{6t} \quad (13)$$

In equation (11) we impose  $\varepsilon_{recov}$ , and thus it obtains  $T_{3t}$ . The equation (12) can be solved numerically by an iterative method and gives  $T_{6t}$ .

- ◆ Specific powers interactions:

$$w_C = - \int_{T_1}^{T_{2r}} c_p \cdot dT \quad (14)$$

$$w_T = - \int_{T_{4r}}^{T_{5r}} c_p \cdot dT \quad (15)$$

$$w_{cycle} = w_T + w_C, w_{supplied} = 0.99 \cdot w_{cycle} \quad (16)$$

- ◆ Specific heat rate interactions:

$$q_{solar} = \int_{T_{3r}}^{T_{4t}} c_p \cdot dT \quad (17)$$

$$q_{cog} = \int_{T_{6r}}^{T_{7r}} c_p \cdot dT \quad (18)$$

$$q_{exhaust} = \int_{T_{7r}}^{T_1} c_p \cdot dT \quad (19)$$

- ◆ First law efficiencies:

$$\eta_w = \frac{w_{supplied}}{q_{solar}} \quad (20)$$

$$\eta_{cog} = \frac{w_{supplied} + q_{cog}}{q_{solar}} \quad (21)$$

### 3. NUMERICAL RESULTS

The numerical results are shown in Figures 2 to 10.

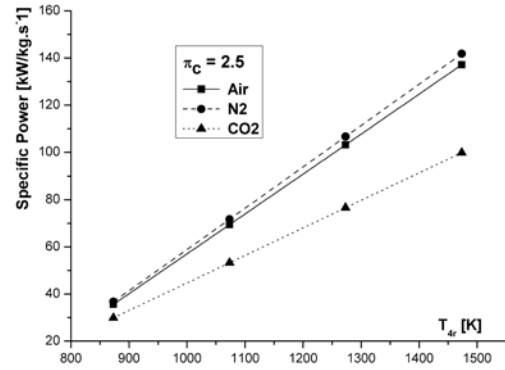


Fig. 2. Specific power versus maximum temperature on the cycle.  $\pi_C = 2.5$

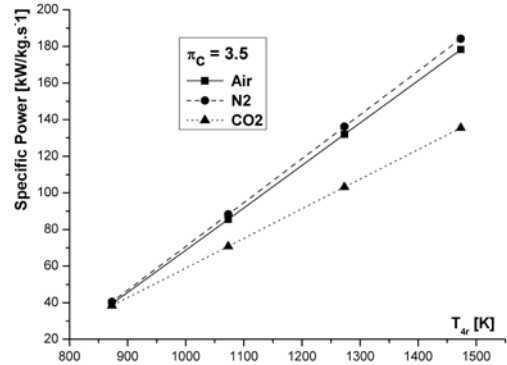


Fig. 3. Specific power versus maximum temperature on the cycle.  $\pi_C = 3.5$

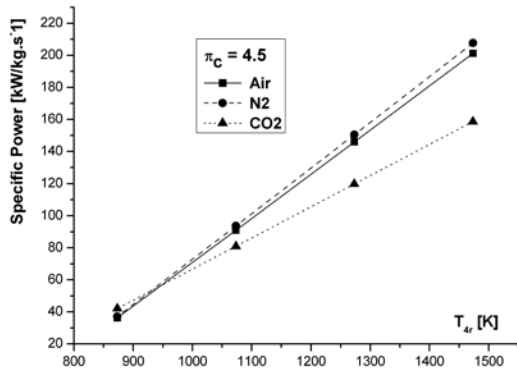


Fig. 4. Specific power versus maximum temperature on the cycle.  $\pi_C = 4.5$

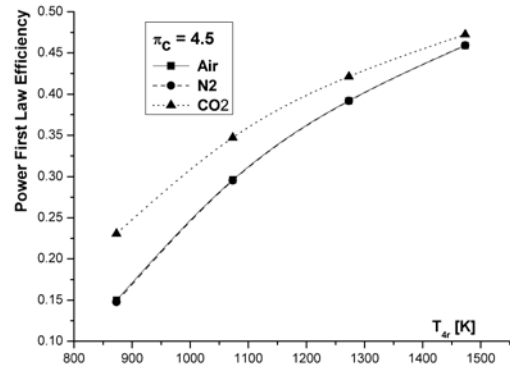


Fig. 7. Power first law efficiency versus maximum temperature on the cycle.  $\pi_C = 4.5$

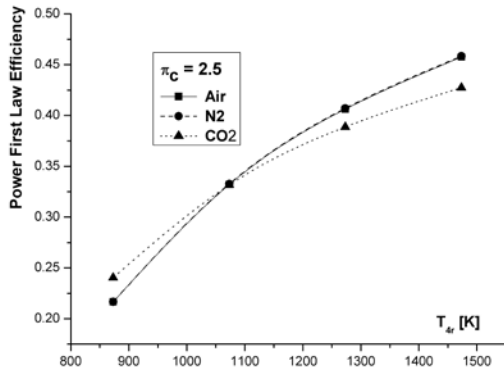


Fig. 5. Power first law efficiency versus maximum temperature on the cycle.  $\pi_C = 2.5$

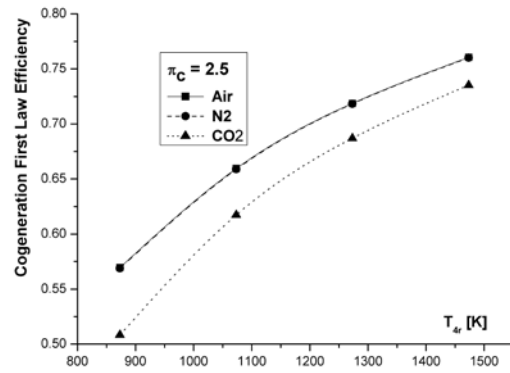


Fig. 8. Cogeneration first law efficiency versus maximum temperature on the cycle.  $\pi_C = 2.5$

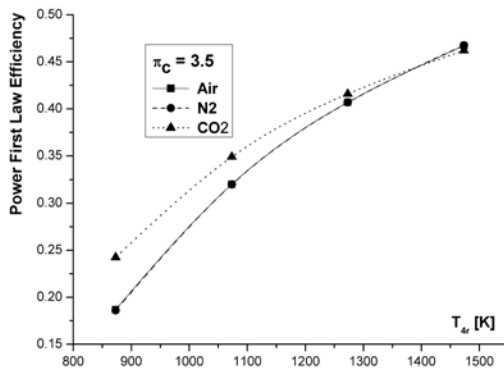


Fig. 6. Power first law efficiency versus maximum temperature on the cycle.  $\pi_C = 3.5$

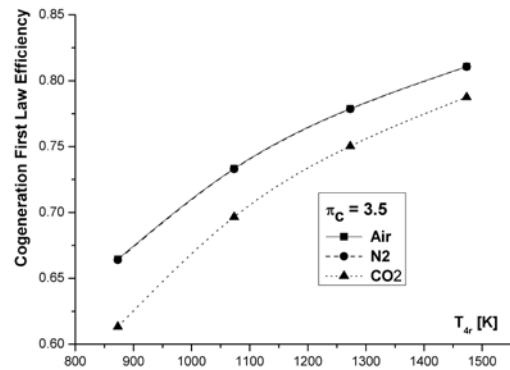


Fig. 9. Cogeneration first law efficiency versus maximum temperature on the cycle.  $\pi_C = 3.5$

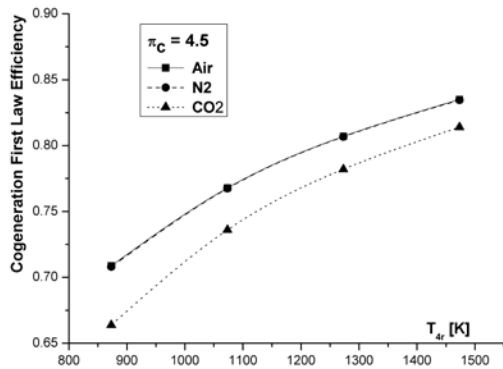


Fig. 10. Cogeneration first law efficiency versus maximum temperature on the cycle.  $\pi_c = 4.5$

#### 4. CONCLUSIONS

The numerical results revealed that the nature of the working fluid is controlling the performances of the solar cogeneration cycle. At a first look, we cannot say which working fluid is more appropriate to solar applications.

The main conclusions emphasize that:

- ❖ the air and nitrogen are quite similar in effects on the specific power and the first law efficiencies,
- ❖ the specific power supplied by the carbon dioxide is smaller than those supplied by air and nitrogen,

- ❖ the power first law efficiencies of carbon dioxide really can compete with those of air/nitrogen, especially at larger compression ratios and at lesser  $T_{4r}$ ,
- ❖ for the carbon dioxide, the cogeneration first law efficiencies are smaller than those for air and nitrogen; the cogeneration first law efficiencies might be actually compared in a specific solar application, when the ratio of cogenerated heat to supplied power is imposed by customers.

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*Muzeul Olteniei din Craiova reunește mai multe institutii muzeale create și dezvoltate de la începutul sec. al XIX-lea și pînă astăzi. Sediul central al Muzeului Olteniei se află în clădirea monumentală, din str. Popa Sapca nr.4, proiectată de cunoscutul arhitect C. Iotu și construită, începînd din anul 1920 de firmele de construcții Dalla Barba și Peressutti. Palatul "Ramuri" cum este cunoscut edificiul se află în centrul municipiului Craiova, între Palatul Jean Mihail, azi Muzeul de Artă, Colegiul Carol I și Biserica Sfînta Treime, ctitorie a familiei princiare Barbu Stirbei și Palatul Prefecturii Judetului Dolj proiectat de arhitect P. Antonescu.*